



# FREE TRANSVERSE VIBRATIONS OF ELASTICALLY CONNECTED SIMPLY SUPPORTED DOUBLE-BEAM COMPLEX SYSTEM

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In this paper, the free vibration analysis of two parallel simply supported beams continuously joined by a Winkler elastic layer is presented. The motion of the system is described by a homogeneous set of two partial differential equations, which is solved by using the classical Bernoulli–Fourier method. The natural frequencies of the system are determined. The initial-value problem is considered to find the final form of the free vibrations. The free vibrations of an elastically connected double-beam complex system are realized by synchronous and asynchronous deflections. The presented theoretical analysis is illustrated by a numerical example, in which the effect of physical parameters characterizing the vibrating system on the natural frequencies is investigated.

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## 1. INTRODUCTION

An elastically connected double-beam system is another model of a complex continuous system consisting of two one-dimensional solids joined by linear, elastic layer of a Winkler type. Analogously built double-string system is the simplest model of a complex continuous system. The transverse vibration analysis of this interesting vibratory system has been presented by the author in a few of his early works [1–9]. In the present paper the free vibrations of a double-beam system are considered for the simple case of simply supported boundary conditions. Assumed particular boundary conditions make it possible to solve the vibration problem for the beams by using the less general mathematical procedures. Because of the same boundary conditions the vibration problems for both of these complex continuous systems discussed (a double-string and double-beam system) are similar. Then the vibration analysis of a double-string system can be helpful in the investigation of a system considered.

The different aspects of dynamics of an elastically connected double-beam system have been treated by many authors: Seelig and Hoppmann II [10, 11], Kessel [12], Kessel and Raske [13], Saito and Chonan [14, 15], Kozlov [16], Kashin [17], Rao [18], Lu and Douglas [19], Douglas and Yang [20], Douglas [21], Oniszczuk [22–31], Chonan [32, 33], Hyer *et al.* [34, 35], Stepanov [36], Dmitriyev [37], Hamada *et al.* [38, 39], Yamaguchi and Saito [40], Joshi and Upadhyaya [41], Sylwan [42], Vaswani *et al.* [43], Kokhmaniuk [44], Yankelevsky [45], Aida *et al.* [46], Frostig and Baruch [47, 48], Kukla and Skalmierski [49], Kukla [50, 51], Macé [52], Chen and Sheu [53, 54], Chen and Lin [55], Lueschen and Bergman [56], Sakiyama *et al.* [57, 58], Szcześniak [59, 60], Kawazoe *et al.* [61], and Cabańska-Płaczkiwicz [62–67]. The works [2, 4, 5, 36, 38–40, 46, 55, 61]

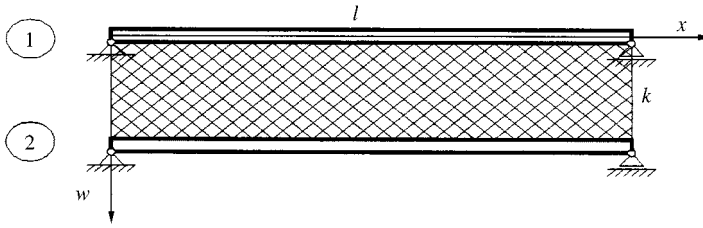


Figure 1. The physical model of an elastically connected double-beam complex system.

devoted to applying a double-beam system as a continuous dynamic vibration absorber (CDVA) are especially interesting because of the great practical importance of CDVAs in many fields of civil and mechanical engineering.

In the present paper, the complete exact theoretical solutions of the free vibrations of simply supported double-beam system are formulated. The general free vibration analysis of an elastically connected double-beam system is complicated and laborious in view of the large variety of possible combinations of the boundary conditions. Therefore, in this work the considerations are limited only to the case of simply supported beams. The vibrations of a general system with the homogeneous boundary conditions will be discussed by the author in the next publications concerning this subject.

## 2. FORMULATION OF THE PROBLEM

The physical model of the vibrating system under consideration is composed of two parallel, slender, prismatic and homogeneous beams joined by a Winkler elastic layer (see Figure 1). Both beams have the same length and are simply supported at their ends. The small undamped vibrations of the system are considered.

Using the Bernoulli–Euler beam theory the free transverse vibrations of an elastically connected double-beam system are described by the following differential equations [2, 24]:

$$K_1 w_1^{IV} + m_1 \ddot{w}_1 + k(w_1 - w_2) = 0, \quad K_2 w_2^{IV} + m_2 \ddot{w}_2 + k(w_2 - w_1) = 0, \quad (1)$$

where  $w_i = w_i(x, t)$  is the transverse beam deflection;  $x, t$  are the spatial co-ordinate and the time;  $E_i$  is the Young modulus of elasticity;  $F_i$  is the cross-sectional area of the beam;  $J_i$  is the moment of inertia of the beam cross-section;  $K_i$  is the flexural rigidity of the beam;  $k$  is the stiffness modulus of a Winkler elastic layer;  $l$  is the length of the beam and  $\rho_i$  is the mass density,

$$K_i = E_i J_i, \quad m_i = \rho_i F_i, \quad \dot{w}_i = \partial w_i / \partial t, \quad w'_i = \partial w_i / \partial x, \quad i = 1, 2.$$

The initial conditions in general form and boundary conditions for simply supported beams are assumed as follows:

$$w_i(0, t) = w''_i(0, t) = w_i(l, t) = w''_i(l, t) = 0, \quad (2)$$

$$w_i(x, 0) = w_{i0}(x), \quad \dot{w}_i(x, 0) = v_{i0}(x), \quad i = 1, 2. \quad (3)$$

## 3. SOLUTION OF THE PROBLEM

The homogeneous partial differential equations (1) with the governing boundary conditions (2) can be solved by the Bernoulli–Fourier method assuming the solutions

in the form

$$w_i(x, t) = \sum_{n=1}^{\infty} X_n(x)S_{in}(t) = \sum_{n=1}^{\infty} \sin(k_n x)S_{in}(t), \quad i = 1, 2, \tag{4}$$

where  $S_{in}(t)$  is the unknown time function;

$$X_n(x) = \sin(k_n x), \quad k_n = l^{-1}n\pi, \quad n = 1, 2, 3, \dots \tag{5}$$

$X_n(x)$  is the known mode shape function for simply supported single beam.

Introducing the general solutions (4) into equations (1) one gets the following relations:

$$\begin{aligned} \sum_{n=1}^{\infty} [\ddot{S}_{1n} + (K_1 k_n^4 + k)m_1^{-1}S_{1n} - km_1^{-1}S_{2n}]X_n &= 0, \\ \sum_{n=1}^{\infty} [\ddot{S}_{2n} + (K_2 k_n^4 + k)m_2^{-1}S_{2n} - km_2^{-1}S_{1n}]X_n &= 0. \end{aligned}$$

From the above one obtains a set of ordinary differential equations for the unknown time functions

$$\ddot{S}_{1n} + \Omega_{11n}^2 S_{1n} - \Omega_{10}^2 S_{2n} = 0, \quad \ddot{S}_{2n} + \Omega_{22n}^2 S_{2n} - \Omega_{20}^2 S_{1n} = 0, \tag{6}$$

where

$$\Omega_{iin}^2 = (K_i k_n^4 + k)m_i^{-1}, \quad \Omega_{i0}^2 = km_i^{-1}, \quad \Omega_{120}^2 = \Omega_{10}^2 \Omega_{20}^2 = k^2(m_1 m_2)^{-1}, \quad i = 1, 2,$$

$\Omega_{iin}$  and  $\Omega_{120}$  denote partial and coupling frequency of the system respectively. The solutions of equations (6) are as follows:

$$S_{1n}(t) = C_n e^{i\omega_n t}, \quad S_{2n}(t) = D_n e^{i\omega_n t}, \quad i = (-1)^{1/2}, \tag{7}$$

where  $\omega_n$  is the natural frequency of the system. By substituting them into equations (6) results in the following system of homogeneous algebraic equations for the unknown constants  $C_n, D_n$ :

$$(\Omega_{11n}^2 - \omega_n^2)C_n - \Omega_{10}^2 D_n = 0, \quad (\Omega_{22n}^2 - \omega_n^2)D_n - \Omega_{20}^2 C_n = 0. \tag{8}$$

Equations (8) have non-trivial solutions in the case when the determinant of the system coefficient matrix is equal to zero. This yields the following frequency (characteristic) equation:

$$\omega_n^4 - (\Omega_{11n}^2 + \Omega_{22n}^2)\omega_n^2 + (\Omega_{11n}^2 \Omega_{22n}^2 - \Omega_{120}^4) = 0 \tag{9}$$

or

$$\begin{aligned} \omega_n^4 - [(K_1 k_n^4 + k)m_1^{-1} + (K_2 k_n^4 + k)m_2^{-1}]\omega_n^2 \\ + k_n^4 [K_1 K_2 k_n^4 + k(K_1 + K_2)](m_1 m_2)^{-1} = 0. \end{aligned} \tag{10}$$

Since the discriminant of this biquadratic algebraic equation is positive

$$D = (\Omega_{11n}^2 + \Omega_{22n}^2)^2 - 4(\Omega_{11n}^2 \Omega_{22n}^2 - \Omega_{120}^4) = (\Omega_{11n}^2 - \Omega_{22n}^2)^2 + 4\Omega_{120}^4 > 0$$

and the relationships mentioned below are also satisfied

$$(\Omega_{11n}^2 \Omega_{22n}^2 - \Omega_{120}^4) > 0, \quad (\Omega_{11n}^2 + \Omega_{22n}^2) > D^{1/2},$$

then the characteristic equation (9) has two different, real, positive roots  $\omega_{1,2n}^2$ :

$$\omega_{1,2n}^2 = 0.5 \{ (\Omega_{11n}^2 + \Omega_{22n}^2) \mp [(\Omega_{11n}^2 - \Omega_{22n}^2)^2 + 4\Omega_{120}^4]^{1/2} \}, \quad \omega_{1n} < \omega_{2n}. \quad (11)$$

One obtains two infinite sequences of natural frequencies  $\omega_{1n}, \omega_{2n}$  ( $\omega_{1n} < \omega_{2n}$ ) in the form

$$\begin{aligned} \omega_{1,2n}^2 = 0.5 \{ & [(K_1 k_n^4 + k) m_1^{-1} + (K_2 k_n^4 + k) m_2^{-1}] \mp [(K_1 k_n^4 + k) m_1^{-1} \\ & + (K_2 k_n^4 + k) m_2^{-1}]^2 - 4k_n^4 (m_1 m_2)^{-1} [K_1 K_2 k_n^4 + k(K_1 + K_2)] \}^{1/2}. \quad (12) \end{aligned}$$

Now the solutions (7) may be written as

$$\begin{aligned} S_{1n}(t) &= C_{1n} e^{i\omega_{1n}t} + C_{2n} e^{-i\omega_{1n}t} + C_{3n} e^{i\omega_{2n}t} + C_{4n} e^{-i\omega_{2n}t}, \\ S_{2n}(t) &= D_{1n} e^{i\omega_{1n}t} + D_{2n} e^{-i\omega_{1n}t} + D_{3n} e^{i\omega_{2n}t} + D_{4n} e^{-i\omega_{2n}t}. \end{aligned}$$

After rearranging the above relations (introducing the trigonometric functions) the unknown time functions are expressed by

$$\begin{aligned} S_{1n}(t) &= \sum_{i=1}^2 T_{in}(t) = \sum_{i=1}^2 [A_{in} \sin(\omega_{in}t) + B_{in} \cos(\omega_{in}t)], \\ S_{2n}(t) &= \sum_{i=1}^2 a_{in} T_{in}(t) = \sum_{i=1}^2 [A_{in} \sin(\omega_{in}t) + B_{in} \cos(\omega_{in}t)] a_{in}, \end{aligned} \quad (13)$$

where

$$T_{in}(t) = A_{in} \sin(\omega_{in}t) + B_{in} \cos(\omega_{in}t), \quad (14)$$

$$\begin{aligned} a_{in} &= (K_1 k_n^4 + k - m_1 \omega_{in}^2) k^{-1} = k(K_2 k_n^4 + k - m_2 \omega_{in}^2)^{-1} = \Omega_{10}^{-2} (\Omega_{11n}^2 - \omega_{in}^2) \\ &= \Omega_{20}^2 (\Omega_{22n}^2 - \omega_{in}^2)^{-1}, \quad k_n = l^{-1} n\pi, \quad i = 1, 2, \quad n = 1, 2, 3, \dots \end{aligned} \quad (15)$$

It is important to note that the coefficients  $a_{in}$  (15) are as follows:

$$\begin{aligned} a_{1,2n} &= 0.5 \Omega_{10}^{-2} \{ (\Omega_{11n}^2 - \Omega_{22n}^2) \pm [(\Omega_{11n}^2 - \Omega_{22n}^2)^2 + 4\Omega_{120}^4]^{1/2} \}, \quad a_{1n} > 0, \quad a_{2n} < 0, \\ a_{1n} a_{2n} &= -m_1 m_2^{-1} = -M_1 M_2^{-1} = -\Omega_{10}^{-2} \Omega_{20}^2, \quad M_i = m_i l = \rho_i F_i l. \end{aligned}$$

It is proved that the coefficient  $a_{1n}$  dependent on lower natural frequency  $\omega_{1n}$  is always positive while  $a_{2n}$  dependent on higher frequency  $\omega_{2n}$  is always negative.

Finally, the free transverse vibrations of an elastically connected simply supported double-beam complex system are described by the following formulae:

$$\begin{aligned}
 w_1(x, t) &= \sum_{n=1}^{\infty} X_n(x)S_{1n}(t) = \sum_{n=1}^{\infty} X_n(x) \sum_{i=1}^2 T_{in}(t) = \sum_{n=1}^{\infty} \sum_{i=1}^2 X_{1in}(x)T_{in}(t) \\
 &= \sum_{n=1}^{\infty} \sin(k_n x) \sum_{i=1}^2 [A_{in} \sin(\omega_{in}t) + B_{in} \cos(\omega_{in}t)],
 \end{aligned}
 \tag{16}$$

$$\begin{aligned}
 w_2(x, t) &= \sum_{n=1}^{\infty} X_n(x)S_{2n}(t) = \sum_{n=1}^{\infty} X_n(x) \sum_{i=1}^2 a_{in} T_{in}(t) = \sum_{n=1}^{\infty} \sum_{i=1}^2 a_{in} X_n(x)T_{in}(t) \\
 &= \sum_{n=1}^{\infty} \sum_{i=1}^2 X_{2in}(x)T_{in}(t) = \sum_{n=1}^{\infty} \sin(k_n x) \sum_{i=1}^2 [A_{in} \sin(\omega_{in}t) + B_{in} \cos(\omega_{in}t)]a_{in},
 \end{aligned}$$

where

$$X_{1in}(x) = X_n(x) = \sin(k_n x), \quad X_{2in}(x) = a_{in}X_n(x) = a_{in} \sin(k_n x).
 \tag{17}$$

The functions  $X_{1in}(x)$ ,  $X_{2in}(x)$  are the natural mode shapes of vibration of a beam system corresponding to two sequences of the natural frequencies  $\omega_{in}$ .

An elastically connected simply supported double-beam complex system executes two kinds of vibrating motions: synchronous vibrations ( $a_{1n} > 0$ ) with lower natural frequencies  $\omega_{1n}$  and asynchronous vibrations ( $a_{2n} < 0$ ) with higher frequencies  $\omega_{2n}$ .

The general mode shapes of vibration shown in Figure 2 are the same as the natural mode shapes determined for a doubling-string system [1, 2, 8]. It is also seen that the nature of the free vibrations for a double-beam system is analogous to that for a double-string system. The mathematical form of the corresponding solutions is identical for both systems as a consequence of governing the same boundary conditions.

To find the final form of the free vibrations the initial-value problem is solved. The unknown constants  $A_{in}$  and  $B_{in}$  are determined from the assumed initial conditions (3) using the orthogonality property of mode shape functions. In this case the classical orthogonality condition is applied:

$$\begin{aligned}
 \int_0^l X_m X_n dx &= \int_0^l \sin(k_m x) \sin(k_n x) dx = c\delta_{mn}, \\
 c = c_n^2 &= \int_0^l X_n^2 dx = \int_0^l \sin^2(k_n x) dx = 0.5l,
 \end{aligned}
 \tag{18}$$

$\delta_{mn}$  is the Kronecker delta function:  $\delta_{mn} = 0$  for  $m \neq n$  and  $\delta_{mn} = 1$  for  $m = n$ . Substitution the solutions (16) into the initial conditions (3) gives

$$\begin{aligned}
 w_{10} &= \sum_{n=1}^{\infty} X_n \sum_{i=1}^2 B_{in}, & v_{10} &= \sum_{n=1}^{\infty} X_n \sum_{i=1}^2 \omega_{in} A_{in}, \\
 w_{20} &= \sum_{n=1}^{\infty} X_n \sum_{i=1}^2 a_{in} B_{in}, & v_{20} &= \sum_{n=1}^{\infty} X_n \sum_{i=1}^2 a_{in} \omega_{in} A_{in}.
 \end{aligned}$$

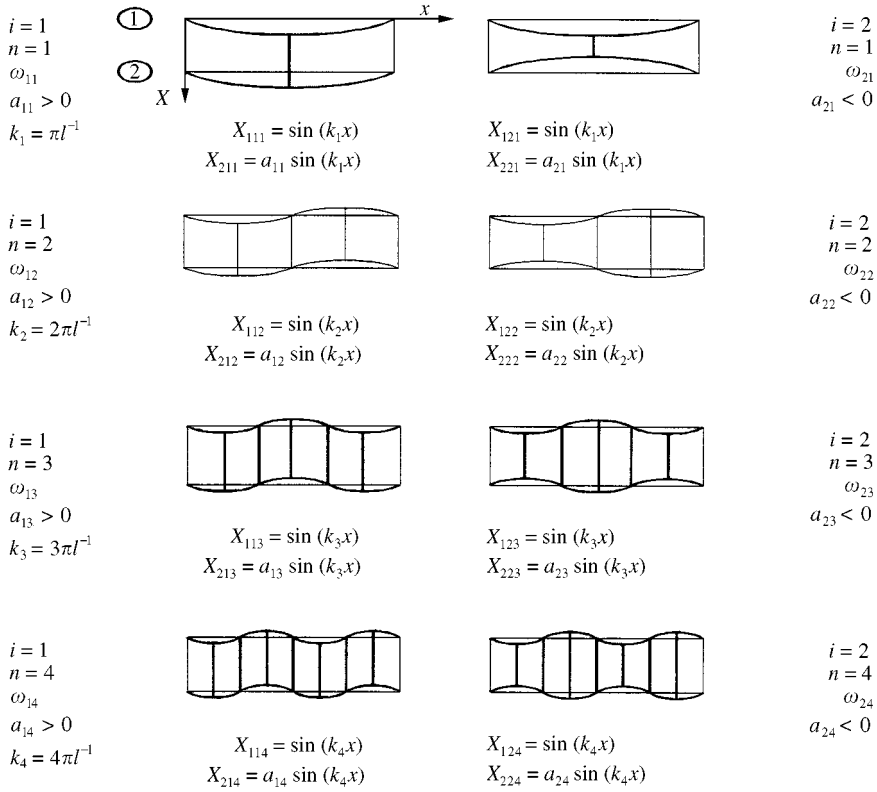


Figure 2. The general mode shapes of vibration of an elastically connected simply supported double-beam complex system corresponding to the first four pairs of the natural frequencies.

Multiplying the above relations by the eigenfunction  $X_m$ , then integrating them with respect to  $x$  from 0 to  $l$  and using the orthogonality condition (18) produces

$$c^{-1} \int_0^l w_{10} X_n dx = \sum_{i=1}^2 B_{in}, \quad c^{-1} \int_0^l v_{10} X_n dx = \sum_{i=1}^2 \omega_{in} A_{in},$$

$$c^{-1} \int_0^l w_{20} X_n dx = \sum_{i=1}^2 a_{in} B_{in}, \quad c^{-1} \int_0^l v_{20} X_n dx = \sum_{i=1}^2 a_{in} \omega_{in} A_{in}.$$

Solving these equations the following formulae making it possible to calculate the unknown constants are obtained:

$$A_{1n} = (c_{1n} \omega_{1n})^{-1} \int_0^l (a_{2n} v_{10} - v_{20}) \sin(k_n x) dx,$$

$$A_{2n} = (c_{2n} \omega_{2n})^{-1} \int_0^l (a_{1n} v_{10} - v_{20}) \sin(k_n x) dx,$$

$$B_{1n} = c_{1n}^{-1} \int_0^l (a_{2n} w_{10} - w_{20}) \sin(k_n x) dx,$$

$$B_{2n} = c_{2n}^{-1} \int_0^l (a_{1n} w_{10} - w_{20}) \sin(k_n x) dx,$$
(19)

where

$$c_{1n} = -c_{2n} = (a_{2n} - a_{1n})c = 0.5l\Omega_{10}^{-2}(\omega_{1n}^2 - \omega_{2n}^2).$$

#### 4. NUMERICAL EXAMPLE

The free transverse vibrations of two simply supported beams are considered to establish the effect of physical parameters characterizing the vibrating system on the natural frequencies.

The following values of the parameters are used in the numerical calculations:

$$E = 1 \times 10^{10} \text{ N m}^{-2}, \quad F = 5 \times 10^{-2} \text{ m}^2, \quad J = 4 \times 10^{-4} \text{ m}^4,$$

$$K = EJ = 4 \times 10^6 \text{ N m}^2, \quad k = (0.5) \times 10^5 \text{ N m}^{-2}, \quad l = 10 \text{ m},$$

$$c = 0.5; 1; 2, \quad m = \rho F = 1 \times 10^2 \text{ kg m}^{-1}, \quad \rho = 2 \times 10^3 \text{ kg m}^{-3}.$$

The problem is solved for three variants of the system which are denoted as

$$\text{V1. } K_1 = K, \quad K_2 = cK, \quad m_1 = m, \quad m_2 = cm,$$

$$\text{V2. } K_1 = K, \quad K_2 = cK, \quad m_i = m, \quad i = 1, 2,$$

$$\text{V3. } K_i = K, \quad i = 1, 2, \quad m_1 = m, \quad m_2 = cm,$$

where  $c$  is a positive constant parameter.

The free vibrations of a double-beam system under consideration are described by formulae (16). The mode shape coefficients  $a_{1n}$  and natural frequencies  $\omega_{in}$  are evaluated from the rearranged expressions (12) and (15) for the above three cases of combinations of the basic parameters of the system

$$\text{V1. } a_{1n} = 1, \quad a_{2n} = -c^{-1}, \quad a_{1n}a_{2n} = -c^{-1},$$

$$\omega_{1n}^2 = Kk_n^4 m^{-1}, \quad \omega_{2n}^2 = \omega_{1n}^2 + \Omega_0^2, \quad \Omega_0^2 = (1 + c^{-1})km^{-1},$$

$$\text{V2. } a_{in} = (Kk_n^4 + k - m\omega_{in}^2)k^{-1}, \quad a_{1n}a_{2n} = -1.$$

$$\omega_{1,2n}^2 = 0.5 \{ [(1+c)Kk_n^4 + 2k]m^{-1} \mp [(1-c)Kk_n^4 m^{-1}]^2 + 4k^2 m^{-2} \}^{1/2},$$

$$\text{V3. } a_{i,n} = (Kk_n^4 + k - m\omega_{in}^2)k^{-1}, \quad a_{1n}a_{2n} = -c^{-1},$$

$$\omega_{1,2n}^2 = 0.5 \{ (1+c^{-1})(Kk_n^4 + k)m^{-1} \mp [(1-c^{-1})(Kk_n^4 + k)m^{-1}]^2 + 4k^2 (cm^2)^{-1} \}^{1/2}.$$

The calculations of  $a_{in}$  and  $\omega_{in}$  are carried out for three values of a parameters  $c$  ( $c = 0.5; 1; 2$ ) as a function of stiffness modulus  $k$ , which is changed in a certain interval  $k = (0.5) \times 10^5 \text{ N m}^{-2}$ . The results of the calculations are presented in Tables 1-9 and

TABLE 1  
Coefficients of natural mode shapes  $a_{in}$

Variant	V1	V1, 2, 3	V1
$c$	0.5	1	2
$a_{1n}$		1	
$a_{2n}$	-2	-1	-0.5

TABLE 2  
Coefficients of natural mode shapes  $a_{in}$

Variant		V2				V3			
$c$		0.5		2		0.5		2	
$k \times 10^{-5}$	$n$	1	2	1	2	1	2	1	2
1	$a_{1n}$	1.10	3.39	0.82	0.16	0.58	0.26	1.25	4.44
	$a_{2n}$	-0.91	-0.29	-1.21	-6.20	-3.50	-7.64	-0.40	-0.11
2	$a_{1n}$	1.05	2.04	0.91	0.30	0.57	0.23	1.13	2.81
	$a_{2n}$	-0.95	-0.49	-1.10	-3.36	-3.52	-8.79	-0.44	-0.18
3	$a_{1n}$	1.03	1.64	0.94	0.40	0.43	0.20	1.09	2.24
	$a_{2n}$	-0.97	-0.61	-1.07	-2.49	-4.68	-9.94	-0.46	-0.22
4	$a_{1n}$	1.02	1.46	0.95	0.49	0.35	0.18	1.06	1.94
	$a_{2n}$	-0.98	-0.69	-1.05	-2.05	-5.77	-10.99	-0.47	-0.26
5	$a_{1n}$	1.02	1.36	0.96	0.55	0.29	0.17	1.05	1.76
	$a_{2n}$	-0.98	-0.74	-1.04	-1.80	-6.86	-12.05	-0.48	-0.28

TABLE 3  
Natural frequencies of double-beam system  $\omega_{in}(s^{-1})$ ; V1;  $c = 0.5$

$k \times 10^{-5}$	$n$	1	2	3	4	5
0	$\omega_{in}$	19.7	79.0	177.7	315.8	493.5
1	$\omega_{2n}$	58.2	96.1	185.9	320.5	496.5
2	$\omega_{2n}$	79.9	110.6	193.8	325.2	499.5
3	$\omega_{2n}$	96.9	123.4	201.4	329.8	502.5
4	$\omega_{2n}$	111.3	135.0	208.7	334.3	505.5
5	$\omega_{2n}$	124.1	145.7	215.8	338.7	508.5

in Figures 3-5. On the diagrams the natural frequencies  $\omega_{in}$  are additionally denoted by subscripts 1, 2, 3, to distinguish the corresponding frequencies computed for  $c = 0.5$ ; 1; 2 respectively. If the natural frequency is independent of constant  $c$ , then this subindex is not applied.

In general, an elastically connected simply supported double-beam system executes two fundamental kinds of vibrations. The system vibrating with lower natural frequencies



TABLE 4

*Natural frequencies of double-beam system  $\omega_{in}(s^{-1})$ ; V1, 2, 3;  $c = 1$*

$k \times 10^{-5}$	$n$	1	2	3	4	5
0	$\omega_{in}$	19.7	79.0	177.7	315.8	493.5
1	$\omega_{2n}$	48.9	90.7	183.2	319.0	495.5
2	$\omega_{2n}$	66.3	101.2	188.6	322.1	497.5
3	$\omega_{2n}$	79.9	110.6	193.8	325.2	499.5
4	$\omega_{2n}$	91.6	119.3	198.9	328.2	501.5
5	$\omega_{2n}$	101.9	127.4	203.9	331.3	503.5

TABLE 5

*Natural frequencies of double-beam system  $\omega_{in}(s^{-1})$ ; V1;  $c = 2$*

$k \times 10^{-5}$	$n$	1	2	3	4	5
0	$\omega_{in}$	19.7	79.0	177.7	315.8	493.5
1	$\omega_{2n}$	43.5	87.9	181.8	318.2	495.0
2	$\omega_{2n}$	58.2	96.1	185.9	320.5	496.5
3	$\omega_{2n}$	69.9	103.6	189.9	322.9	498.0
4	$\omega_{2n}$	79.9	110.6	193.8	325.2	499.5
5	$\omega_{2n}$	88.8	117.2	197.6	327.5	501.0

TABLE 6

*Natural frequencies of double-beam system  $\omega_{in}(s^{-1})$ ; V2;  $c = 0.5$*

$k \times 10^{-5}$	$n$	1	2	3	4	5
0	$\omega_{1n}$	14.0	56.0	125.9	223.9	349.8
	$\omega_{2n}$	19.1	78.8	177.3	315.3	492.6
1	$\omega_{1n}$	17.0	62.0	129.6	226.0	351.2
	$\omega_{2n}$	47.9	86.6	180.3	316.9	493.6
2	$\omega_{1n}$	17.0	64.4	132.7	228.1	352.6
	$\omega_{2n}$	65.5	95.9	183.6	318.5	494.7
3	$\omega_{1n}$	17.0	65.6	135.3	230.1	353.9
	$\omega_{2n}$	79.3	105.1	187.1	320.3	495.7
4	$\omega_{1n}$	17.1	66.3	137.5	232.0	355.3
	$\omega_{2n}$	91.1	113.8	190.8	322.0	496.8
5	$\omega_{1n}$	17.1	66.7	139.3	233.7	356.6
	$\omega_{2n}$	101.5	122.1	194.7	323.9	497.9

$\omega_{1n}$  performs the synchronous vibrations for which the coefficients of mode shapes  $a_{1n}$  are always positive ( $a_{1n} > 0$ ). Next, the asynchronous vibrations are executed with higher frequencies  $\omega_{2n}$  at the mode shape coefficients  $a_{2n}$ , which are always negative ( $a_{2n} < 0$ ).

It is seen that in the case  $c = 1$  all three variants correspond with the simple system of two physically and geometrically identical beams. For this interesting double-beam

TABLE 7  
*Natural frequencies of double-beam system  $\omega_{in}(s^{-1})$ ; V2;  $c = 2$*

$k \times 10^{-5}$	$n$	1	2	3	4	5
0	$\omega_{1n}$	19.7	78.9	177.6	315.7	493.2
	$\omega_{2n}$	27.9	111.7	251.3	446.8	698.1
1	$\omega_{1n}$	23.8	84.1	180.3	317.2	494.2
	$\omega_{2n}$	51.0	116.8	253.4	447.9	698.8
2	$\omega_{1n}$	24.0	87.4	182.8	318.7	495.2
	$\omega_{2n}$	67.8	122.7	255.5	449.1	699.6
3	$\omega_{1n}$	24.0	89.6	185.1	320.2	496.2
	$\omega_{2n}$	81.2	129.2	257.8	450.2	700.3
4	$\omega_{1n}$	24.1	91.0	187.2	321.7	497.2
	$\omega_{2n}$	92.7	135.8	260.0	451.4	701.0
5	$\omega_{1n}$	24.1	92.0	189.1	323.1	498.2
	$\omega_{2n}$	102.9	142.3	262.6	452.6	701.8

TABLE 8  
*Natural frequencies of double-beam system  $\omega_{in}(s^{-1})$ ; V3;  $c = 0.5$*

$k \times 10^{-5}$	$n$	1	2	3	4	5
0	$\omega_{1n}$	19.7	79.0	177.7	315.8	493.5
	$\omega_{2n}$	27.9	111.7	251.2	446.6	697.9
1	$\omega_{1n}$	28.6	83.5	180.3	317.4	494.5
	$\omega_{2n}$	60.5	121.4	255.3	448.9	699.3
2	$\omega_{1n}$	35.4	88.2	182.9	318.9	495.5
	$\omega_{2n}$	76.9	130.1	259.3	451.1	700.8
3	$\omega_{1n}$	45.9	92.9	185.4	320.5	496.5
	$\omega_{2n}$	89.8	138.1	263.2	453.4	702.2
4	$\omega_{1n}$	54.8	97.5	188.0	322.0	497.5
	$\omega_{2n}$	100.8	145.6	267.1	455.6	703.6
5	$\omega_{1n}$	62.7	102.0	190.5	323.5	498.5
	$\omega_{2n}$	110.6	152.6	270.9	457.8	705.0

system one has

$$\omega_{1n}^2 = K k_n^4 m^{-1}, \quad \omega_{2n}^2 = \omega_{1n}^2 + \Omega_0^2, \quad \Omega_0^2 = 2k m^{-1}, \quad a_{1n} = 1, \quad a_{2n} = -1.$$

The important conclusions can be drawn from the above expressions. The synchronous natural frequencies  $\omega_{1n}$  are not dependent on the stiffness modulus  $k$  unlike  $\omega_{2n}$ . The synchronous vibrations are performed by both beams with equal amplitudes ( $a_{1n} = 1$ ), and the lower natural frequencies  $\omega_{1n}$  are the same as for a single beam. As a consequence of this an elastic layer is not deformed on the transverse direction. The asynchronous vibrations are also performed with equal amplitudes ( $a_{2n} = -1$ ), and the natural frequencies  $\omega_{2n}$  are

TABLE 9

Natural frequencies of double-beam system  $\omega_{in}(s^{-1})$ ; V3;  $c = 2$

$k \times 10^{-5}$	$n$	1	2	3	4	5
0	$\omega_{1n}$	14.0	55.8	125.6	223.3	348.9
	$\omega_{2n}$	19.8	79.0	177.8	316.1	493.9
1	$\omega_{1n}$	11.8	52.9	123.8	222.2	348.2
	$\omega_{2n}$	44.1	89.8	183.2	319.2	495.9
2	$\omega_{1n}$	11.6	51.5	122.4	221.2	347.5
	$\omega_{2n}$	58.7	98.8	188.2	322.2	497.9
3	$\omega_{1n}$	11.6	50.2	121.1	220.3	346.8
	$\omega_{2n}$	70.4	106.5	192.9	325.2	499.8
4	$\omega_{1n}$	11.5	49.8	120.1	219.4	346.2
	$\omega_{2n}$	80.3	113.5	197.4	328.1	501.8
5	$\omega_{1n}$	11.5	49.3	119.1	218.6	345.6
	$\omega_{2n}$	89.2	120.1	201.7	330.9	503.7

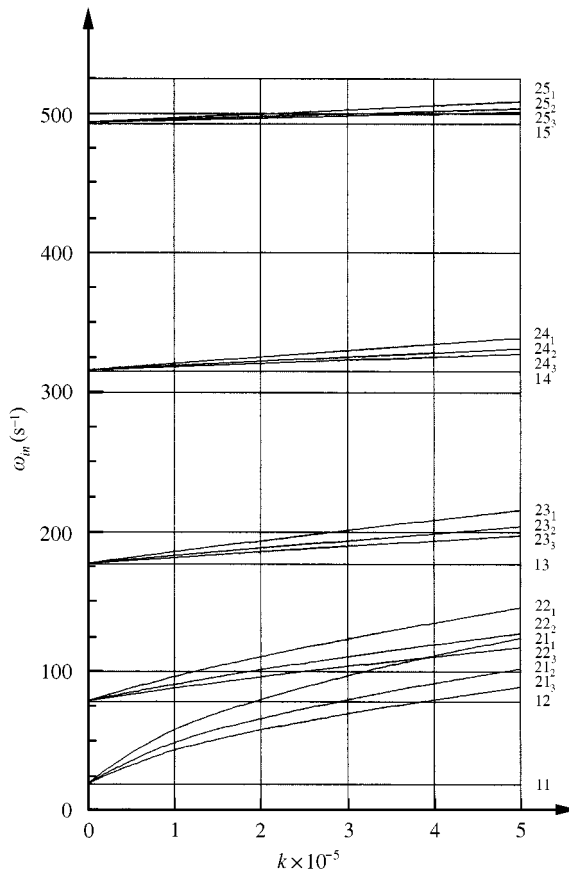


Figure 3. Natural frequencies of double-beam system  $\omega_{in}$  as a function of elastic layer stiffness modulus  $k$ , for various values of a parameter  $c$ ; variant V1. Notation:  $in_1, in_2, in_3$  denote the graphs of  $\omega_{in}$  for  $c = 0.5; 1; 2$  respectively ( $i = 1, 2; n = 1, 2, \dots, 5$ ).

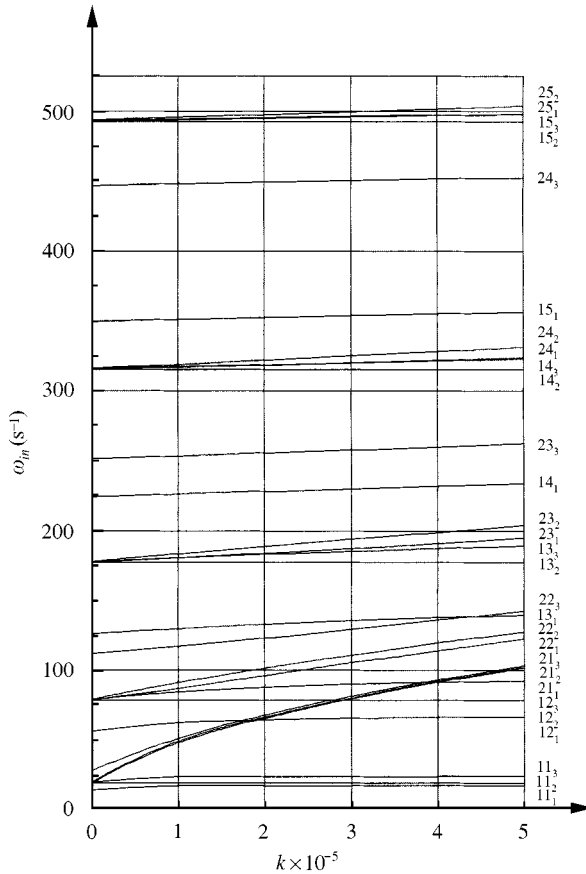


Figure 4. Natural frequencies of double-beam system  $\omega_{in}$  as a function of elastic layer stiffness modulus  $k$ , for various values of a parameter  $c$ ; variant V2. Key as for Figure 3.

identical as for a single beam vibrating on an elastic foundation of stiffness modulus  $2k$ . The exemplary mode shapes of the free vibrations for this particular case are shown in Figure 6.

From the numerical analysis it is seen that there is a general tendency to increase the natural frequencies  $\omega_{in}$  in the case of increasing the layer stiffness modulus  $k$  for each system variant. The increase for lower natural frequencies is greater than for higher ones. For the variant V1 the simultaneous proportional variation of flexural rigidity and mass of the second beam, implies that the synchronous quantities  $a_{1n}$  and  $\omega_{1n}$  are not dependent on an assumed constant  $c$  and layer stiffness modulus  $k$  unlike the asynchronous quantities  $a_{2n}$  and  $\omega_{2n}$ . Their values diminish when a parameter  $c$  grows. For the variant V2 the beams have the same masses and different flexural rigidities. In general, the change of the second beam flexural rigidity causes the natural frequencies to increase, especially the asynchronous frequencies  $\omega_{2n}$  for lower harmonics. For the variant V3 the beams have the same flexural rigidities and different masses. The increase of the second beam mass generates the evident reduction of the natural frequencies which is more considerable for the higher harmonics. This effect is greater for the frequencies  $\omega_{2n}$ .

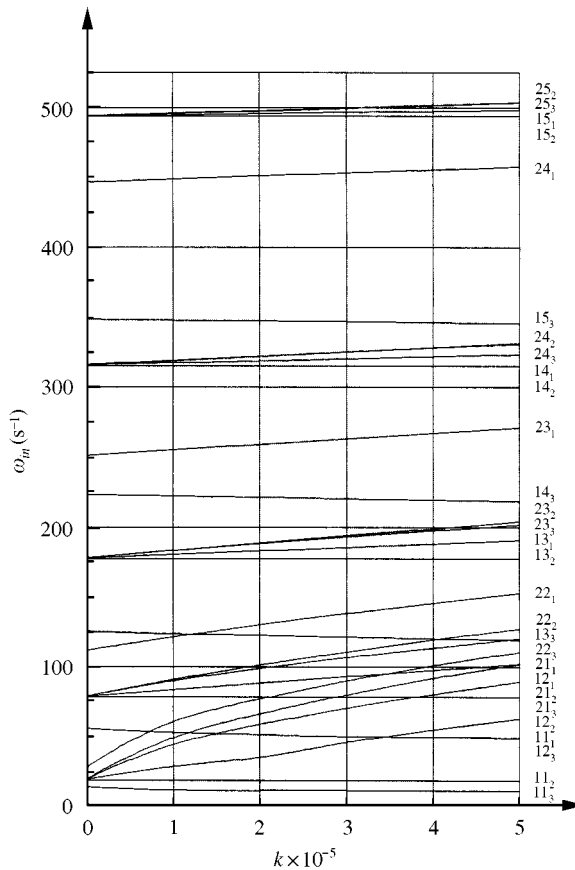


Figure 5. Natural frequencies of double-beam system  $\omega_{in}$  as a function of elastic layer stiffness modulus  $k$ , for various values of a parameter  $c$ ; variant  $V3$ . Key as for Figure 3.

### 5. CONCLUSIONS

The free transverse vibration theory of an elastically connected simply supported double-beam complex system is developed. The solutions of the differential equations of motion are formulated by the classical Bernoulli–Fourier method. The initial-value problem is considered to find the final form of the free vibrations. Two infinite sequences of the natural frequencies  $\omega_{1n}, \omega_{2n}$  ( $\omega_{1n} < \omega_{2n}$ ) are determined. The free vibrations of a double beam are realized by two kinds of motions: synchronous vibrations ( $a_{1n} > 0$ ) with lower natural frequencies  $\omega_{1n}$  and asynchronous vibrations ( $a_{2n} < 0$ ) with higher frequencies  $\omega_{2n}$ . The numerical analysis shows the effect of physical parameters of the system on the natural frequencies. It is seen that the nature of the free vibrations for a simply supported double-beam system and for a double-string system [1, 2, 8] is similar. It can be also shown that a corresponding two-degree-of-freedom complex discrete system described in references [1, 2, 9] is an analogue of an elastically connected double-body complex continuous system represented for example by a double-beam system [1, 38, 39, 46, 55, 61], double-string system [1, 2, 8, 9] and double-membrane system [2, 68]. A beam supported on an elastic foundation is a particular case of a double-beam system considered. The solution procedure applied in this paper can be used for the investigation of general elastically connected multi-beam complex system [2, 18, 24, 45].

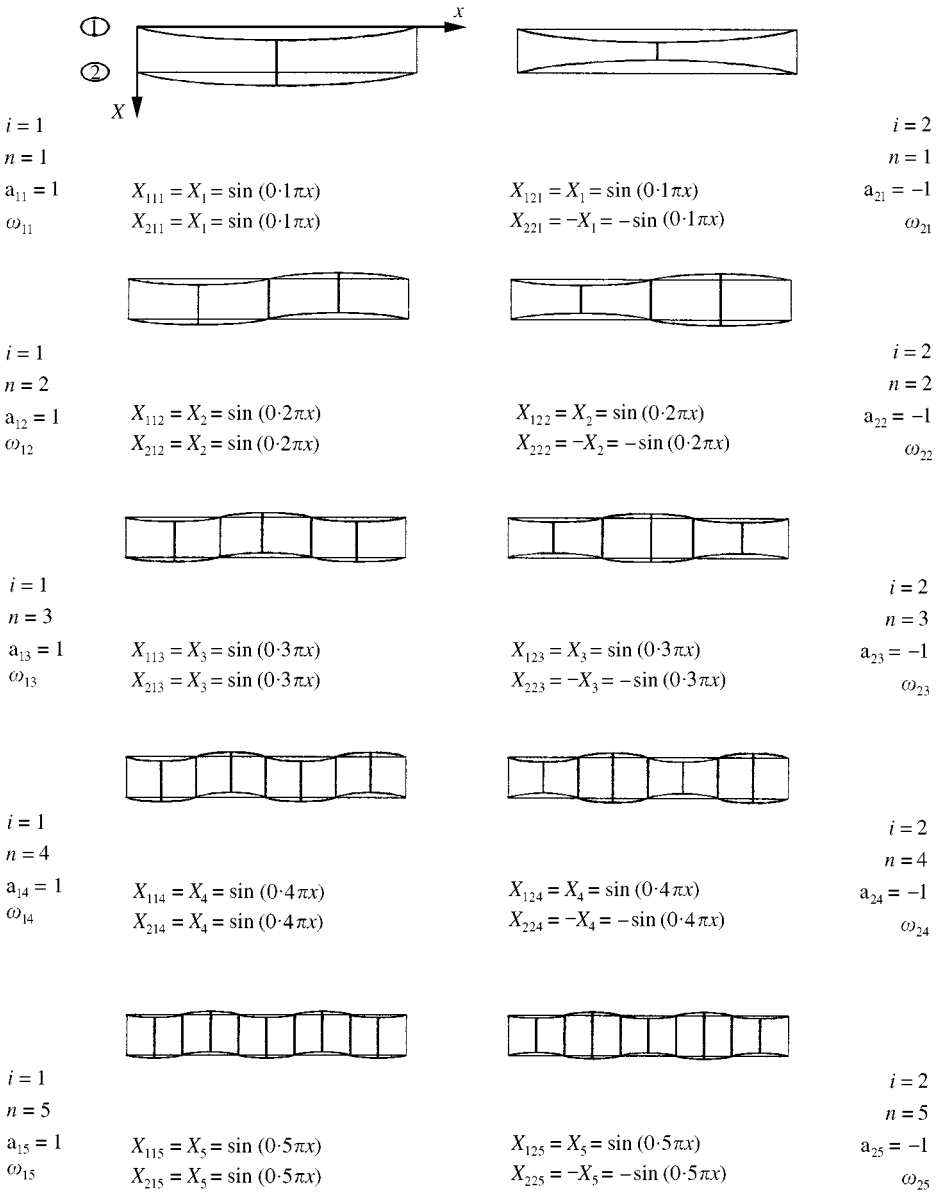


Figure 6. The mode shapes of vibration of an elastically connected simply supported double-beam system corresponding to the first five pairs of the natural frequencies (case  $c = 1$ ).

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